



# Aerodynamic Characteristics of Overtaking Bus under Crosswind: CFD Investigation

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## ABSTRACT

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The presence of other vehicles, travelling together on the road, highly contributes to the condition on the realistic air flow direction. The position alteration of other vehicles generates different airflow direction which is unpredictable experienced by the vehicle under investigation. It considerably affects the stability of a vehicle having large size of the body. The purpose of the study was to investigate the aerodynamic response of a detailed bus model in overtaking process with the interference of crosswind. Fluidic simulation was performed to investigate the air flow behavior imposed on the bus by means of Computational Fluid Dynamics (CFD) analysis. A scaled bus model was used in the simulation with the different positions representing the buses during the overtaking process. The coefficient  $X/L$  was introduced to realize the vehicle position during the overtaking process. The results discussed the alteration of drag force coefficient, lift force coefficient, side force coefficient, during the position on overtaking process. The resulting turbulence kinetic energy around the bus was also discussed by comparing the case without crosswind and when the yaw angle was  $30^\circ$ . The most prominent aerodynamic forces alteration occurs when the overtaking process was at  $X/L=1$ . Further explanation about the pressure coefficient at the surface of the bus and the area around the vehicle was investigated at this position. The accuracy of numerical results calculation was verified by comparing the result of simulation and experimental testing of  $C_d$  and  $C_l$  with the percentage of deviation 0.37% and 2.90% respectively.

## 1. Introduction

The airflow over a road vehicle generates a complex flow field resulted from the varying body shape of the vehicles running on the road and driving conditions including crosswind. The flow becomes more intricate when two road vehicles are in the overtaking process. As a result, the aerodynamic forces imposed on the vehicles vary as the position of the vehicles during the overtaking process changes which affects the vehicle's stability and safety [1-6]. Moreover, a crosswind causes problems for the vehicle's motion and drivability since there are additional lateral air velocity

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components [7]. This issue becomes more severe for a large vehicle such as an intercity bus [8-10].

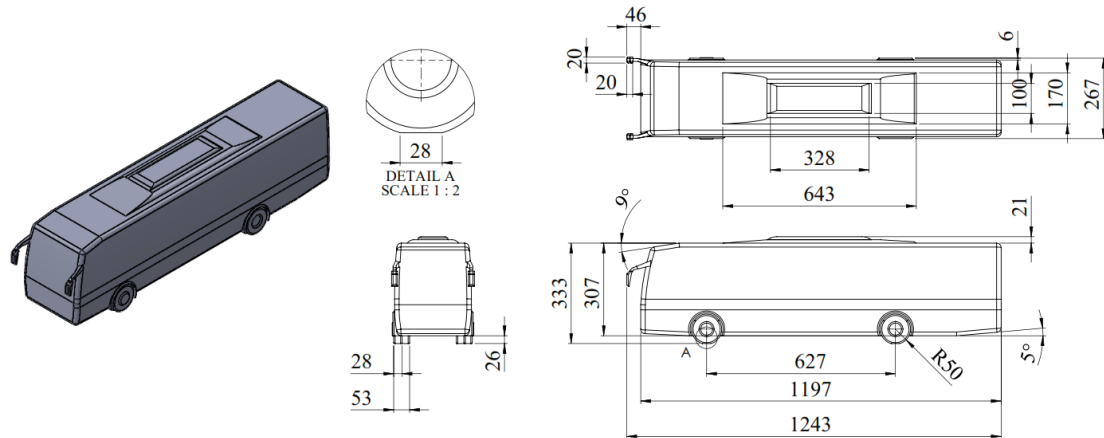
Some studies to investigate the aerodynamic behavior on different driving conditions of large vehicles such as a bus, truck, and trailer have been conducted to study, optimize and experimentally test the aerodynamic characteristics of the vehicle [11-15]. There are also numerous existing numerical and experimental research to investigate the effect of crosswind on a double-deck bus, moving vehicle under crosswind, vulnerability assessment of crosswind and even on passenger trains [16-23]. Their results indicate that generally the side force coefficient of the vehicle due to the presence of crosswind increases as the lateral velocity component due to crosswind elevates. Crosswind itself is not the only factor increasing the uneven airflow condition. However, the overtaking process of vehicles also plays an essential contribution to the instability of the vehicle. Some scholars investigate overtaking process either in simulation or by experimental testing [8,24-26]. Some other researchers also examine the combination between overtaking processes along with the presence of crosswind [27,28]. The investigation includes the effects of aerodynamic force coefficients and the visualization of the flow field around the vehicle.

Most of the mentioned studies were mainly concerned about the effect of crosswind on the aerodynamic characteristics of a simple vehicle model. However, the effects of detailed bus model on the aerodynamic characteristics with the case of overtaking position with another bus by considering asymmetric flow condition is not fully covered and requires a further investigation. This study focused on the investigation of both the overtaking process along with the absence and the presence of crosswind on a detailed intercity bus model with the proposed three-dimensional numerical simulation method. Both aerodynamic characteristics of overtaking and overtaken bus including aerodynamic forces coefficient, pressure coefficient, and resulting turbulence kinetic energy are numerically and visually analyzed.

## **2. Methodology**

### **2.1 Bus Geometry Definition**

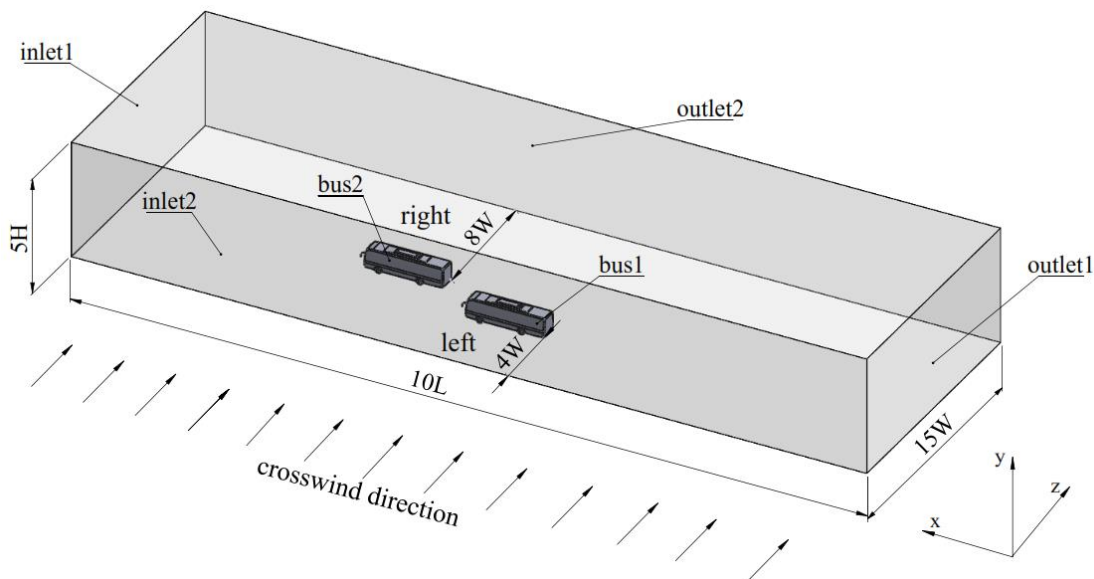
The study used a 1:10 scaled bus model with the dimension is depicted in Figure 1. The length (L) of the bus was 1243 mm, the width (W) was 267 mm, and the overall height (H) was 333 mm. The model used a wheel radius (R) of 50 mm. The assumption of a contact point between the tire and the road was flat due to the elasticity of the tire and the weight of the bus as shown in Detail A in Figure 1. The distance between the front and rear wheel was 627 mm followed by a diffuser with an angle of 5 degrees towards the rear of the bus model. The intersection of the front shield and the upper part of the bus is considered a 9-degree angle slope. The upper part also modeled a compartment where usually an air conditioner equipment was located. The model was also equipped with a pair of mirrors with the dimension was depicted in Figure 1. The door handle, however, was eliminated for the sake of simplicity but keeping the calculation accuracy and the lower part of the bus model was also assumed as flat ignoring the shape of small parts at the underbody. The same assumption applied at the side part of the model disregarding the outer shape of the bus windows.



**Fig. 1.** A detailed intercity bus model (dimensions in mm)

### 2.2 Computational Domain

A cuboid periphery of the computational domain was employed with the overall length  $10L$ , overall width  $15W$ , and overall height  $5H$ . The bus was overtaking on the right side of the overtaken bus so that bus2 represented the overtaking bus and bus1 was the overtaken one. The computational domain utilized two inlet faces and two outlet faces namely inlet1, inlet2, outlet1, and outlet2 as explained in Figure 2. The lower part of the cuboid surface was set as a slip wall represents the moving road surface. The distance between the inlet 1 and the front part of the bus varied depending on the overtaking position yet keeping  $4W$  as the distance between the inlet2 face and the left side of the bus1 model and  $8W$  as the distance between bus2 model and the outlet2 face of the computational domain.



**Fig. 2.** Computational domain of the overtaking bus

### 2.3 Mesh Technique

The mesh generation adopted a hybrid mesh method, utilizing poly-hexcore volume mesh. The smooth transition with three layers was set up at boundary layer settings on fluid walls. To ensure the quality of the calculation, the minimum size of the element was determined at 0.011 m and the

maximum size was 0.30 m with a 1.2 growth rate. Moreover, to efficiently capture the complex shape of the bus model and effectively create the mesh, the local sizing around the area of Bus1 and Bus2 was determined by setting the local min size also 0.01. This setting allows the resulting mesh to generate the mesh results as closely similar as possible to the original model in reality as shown in Figure 3. The resulting mesh produces 812,657 nodes, 6,415 edges and 1,261,389 faces. It can be observed that the generated mesh already represented the real shape of the model and high potential to generate accurate calculation results.

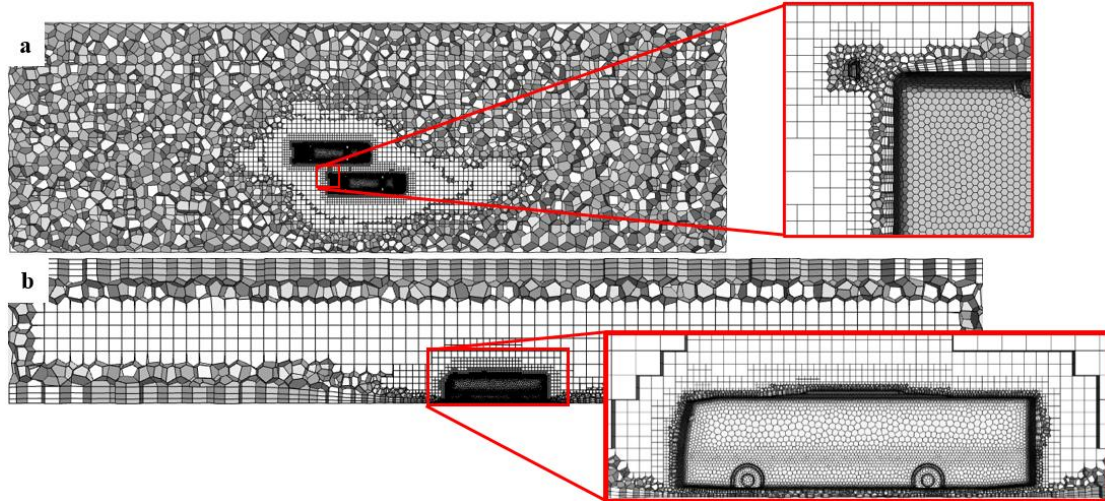


Fig. 3. Mesh strategy during the overtaking position

#### 2.4 Bus Position and Airflow Direction

The typical velocity and natural direction of crosswind vary irregularly over time. However, it was assumed that the velocity and direction of crosswind were constant in every overtaking position and the direction was perpendicular to the bus motion. Instead of keeping two buses in motion, the simulation kept the bus position unchanged and set the velocity of the incoming flow and the crosswind to flow over the bus model. The velocity of incoming flow 10 m/s ( $v_{if}$ ) flowed through the bus model in the -x direction. To simulate the bus2 that was overtaking another bus, the surface of bus2 was set having relative velocity ( $v_{rel}$ ) 3 m/s at the direction of +x. By doing so, this means that bus1 run 10 m/s and was overtaken by bus2 having a velocity of 13 m/s. The crosswind velocity ( $v_{cw}$ ) flowed from the side of the bus perpendicular to the bus motion with a direction of +z. Therefore, a yaw angle ( $\alpha$ ) was produced and defined as the angle between the centerline of the bus and the direction of the resulting velocity ( $v_{res}$ ). Since there were three cases applied to the simulation in this study, there were also three different crosswind velocities for different yaw angles. The velocity of incoming flow was kept as a constant value of 10 m/s and there are three different yaw angles at  $\alpha = 10^\circ$ ,  $\alpha = 20^\circ$ , and  $\alpha = 30^\circ$ , with the crosswind velocity was respectively set at 1.76 m/s, 3.64 m/s, and 5.77 m/s.

The resultant velocity ( $v_{res}$ ) was defined as the vehicle's velocity relative to the direction of the air stream flowing through the Bus1 as follows.

$$v_{res} = \sqrt{v_{if}^2 + v_{cw}^2} \quad (1)$$

$$\alpha = \arctan \left( \frac{v_{cw}}{v_{if}} \right) \quad (2)$$

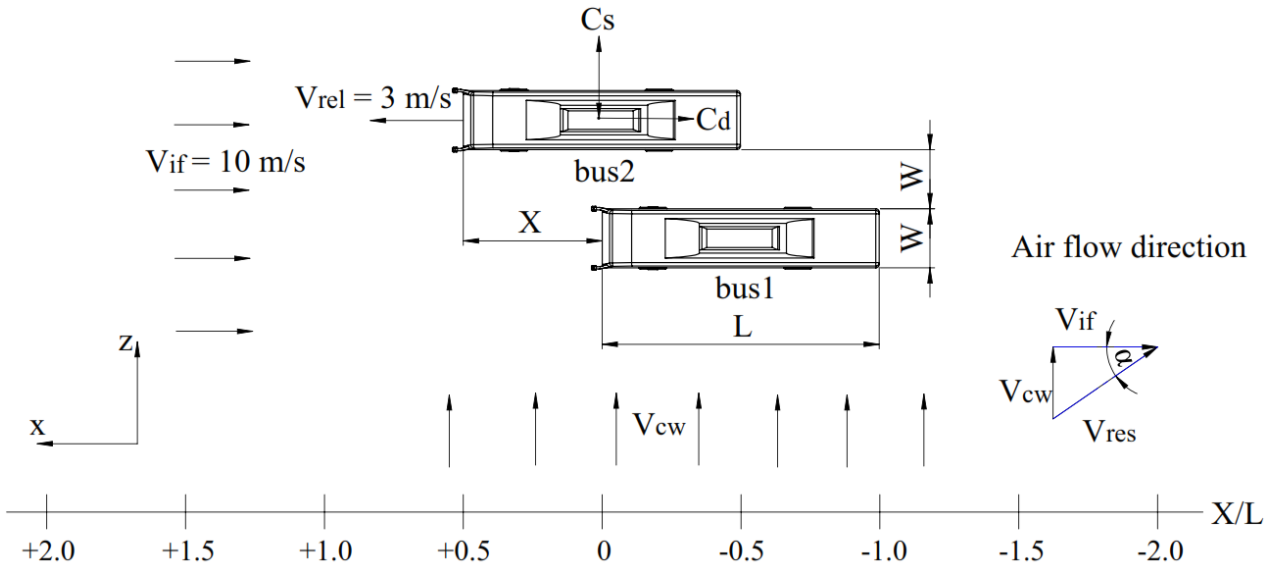


Fig. 4. Bus position and wind direction

Therefore, it can be noted that the vector summation of  $v_{if}$ ,  $v_{rel}$ , and  $v_{cw}$  generates the resultant velocity ( $v_{res}$ ) for Bus2 as follows.

$$v_{res} = \sqrt{\left( v_{if} + (-v_{rel}) \right)^2 + v_{cw}^2} \quad (3)$$

Furthermore, the equation for the yaw angle of Bus2 is:

$$\alpha = \arctan \left( \frac{v_{cw}}{v_{if} + (-v_r)} \right) \quad (4)$$

At this point, it is clear that there is a difference between the yaw angle of Bus1 and Bus2. Therefore, the yaw angle of Bus1 was used as the reference yaw angle of all simulations and results in the discussion. The X/L coefficient was introduced to represent the relative displacement between two buses during the process of overtaking. The X represents the longitudinal distance between Bus1 and Bus2 measured from the front part of the bus. The L value is the length of the bus model. In this study, the corresponding distance during the overtaking process varied from X/L = -2 to X/L = 2. The following discussion divides the X/L in every 0.5.

## 2.5 Operating and Boundary Conditions

The SST k- $\omega$  turbulence model was used in the simulation. The fluid was set to constant density air with the value of 1.225 kg/m<sup>3</sup> with the temperature 288.16°K. This indicates that the air pressure was 1.01325x10<sup>5</sup> Pa. The viscosity was 1.7894x10<sup>-5</sup> kg/m-s. Table 1 shows the setting of boundary conditions.

**Table 1**  
Settings of boundary conditions in the simulation

No	Boundary condition	Settings
1	Inlet1	Velocity-inlet: $v_x = v_{if}, v_y = v_{cw}, v_z = 0$
2	Inlet2	Velocity-inlet: $v_x = v_{if}, v_y = v_{cw}, v_z = 0$
3	Outlet1	Pressure-outlet: atmospheric pressure
4	Outlet2	Pressure-outlet: atmospheric pressure
5	Top	Slip wall
6	Road surface	Slip wall
7	Bus surface	No-slip wall

## 2.6 Aerodynamic Coefficient Definition

The process of a bus overtaking another bus involved aerodynamic forces and pressure that were expressed as non-dimensional coefficients. Three force coefficients namely drag force coefficient ( $C_D$ ), lift force coefficient ( $C_L$ ), and side force coefficient ( $C_S$ ) also one pressure coefficient ( $C_p$ ) are expressed as follows:

$$C_D = \frac{F_D}{\frac{1}{2}\rho v_{res}^2 A} \quad (5)$$

$$C_S = \frac{F_S}{\frac{1}{2}\rho v_{res}^2 A} \quad (6)$$

$$C_L = \frac{F_L}{\frac{1}{2}\rho v_{res}^2 A} \quad (7)$$

$$C_p = \frac{p - p_{if}}{\frac{1}{2}\rho v_{if}^2} \quad (8)$$

where  $F_D$  is the drag force,  $F_S$  is side force, and  $F_L$  is lift force. The  $\rho$  represents the density of the air flowing through the bus model.  $A$  denotes the projection area of the bus.  $v_{res}$  indicates the resultant velocity. The resultant velocity of Bus1 depends on the vector resultants of  $v_{if}$  and  $v_{cw}$ , while Bus2 was determined by the resultant vector of  $v_{if}$ ,  $v_{rel}$  and  $v_{cw}$ . The pressure around the bus was also expressed by a dimensionless coefficient.  $p$  denotes as static pressure in which the area is evaluated,  $p_{if}$  is the static pressure in the free stream,  $\rho$  is the density of the air and  $v_{if}$  is the incoming flow velocity.

## 3. Results

The discussion of the results focusing on aerodynamic force coefficients, turbulence kinetic energy, flow streamlines, and pressure coefficient are investigated as follows.

### 3.1 Effect on $C_d$ , $C_l$ , and $C_s$

Figure 5 shows the effect of bus overtaking position on the performance of aerodynamic forces coefficient on overtaking and overtaken bus at different yaw angles. It can be noted that with the higher magnitude of yaw angle, the value of nearly all aerodynamic coefficients elevates, especially the side force coefficient. This result matches the previous studies by Zhang *et al.*, [15], Dorigatti *et*

*al.*, [21], and Suzuki *et al.*, [29]. As it is compared in Figure 5(a) and Figure 5(b), the drag coefficient of two buses hovers at the value of 0.42 with the absence of crosswind. However, the value of  $C_D$  elevates as the yaw angle increases. As for Bus1, the highest value of  $C_D$  occurs when the yaw angle is  $30^\circ$  at the position of  $X/L$  between 0 and 0.5. It then goes down to the value of around 0.82 when  $X/L$  equals to 1. Other values of  $C_D$  remains unchanged for lower yaw angle ( $\leq 20^\circ$ ). Striking results can be observed for Bus2. The drag coefficient of bus 2 relatively similar between 0.43 – 0.65 at all yaw angle from  $X/L = -2$  to  $X/L = 0.5$ . It then jumps to the value around 1.2 when  $X/L = 1$  at  $30^\circ$  yaw angle. The following results of  $C_D$  go down to the value 0.8.

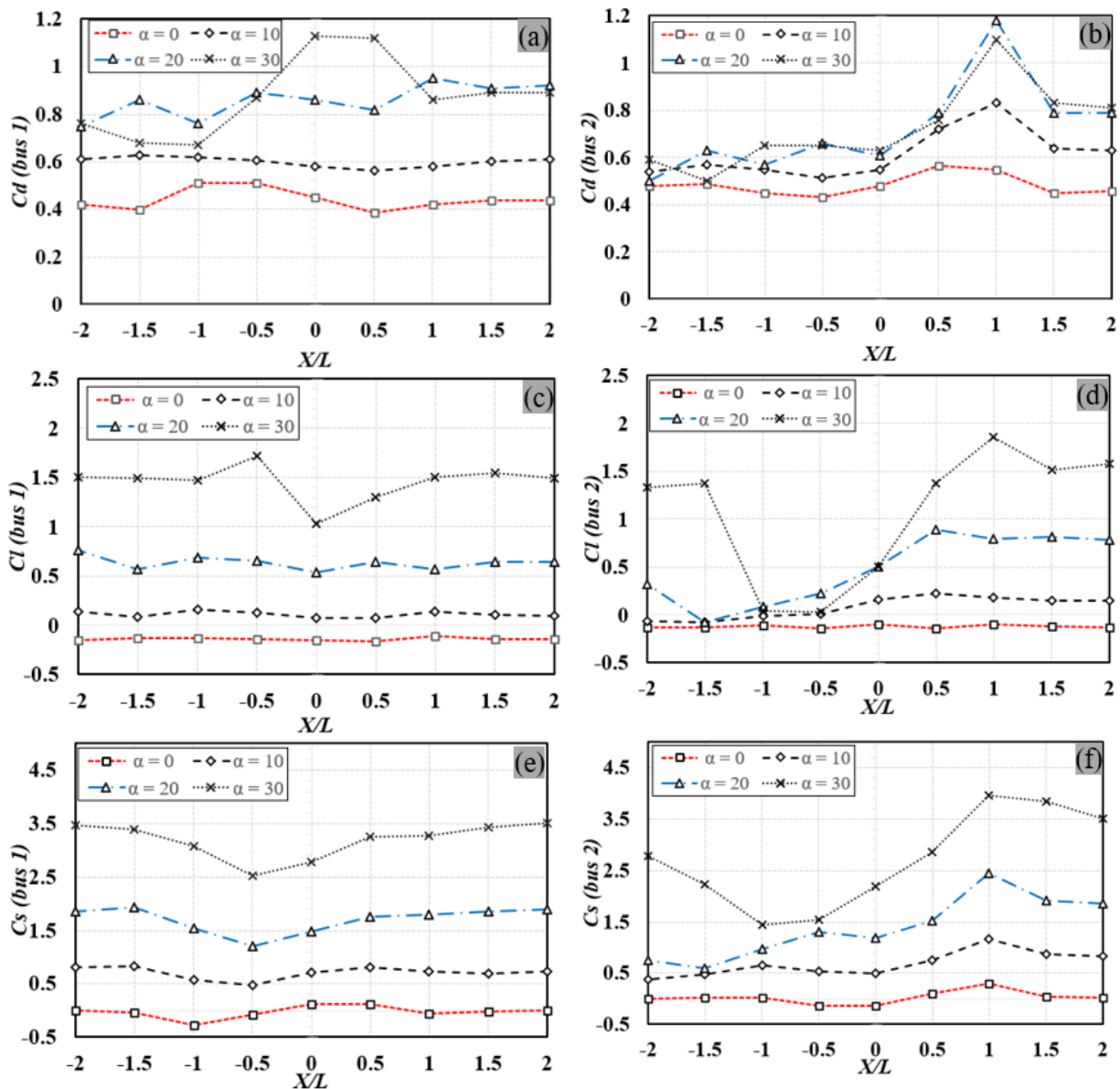


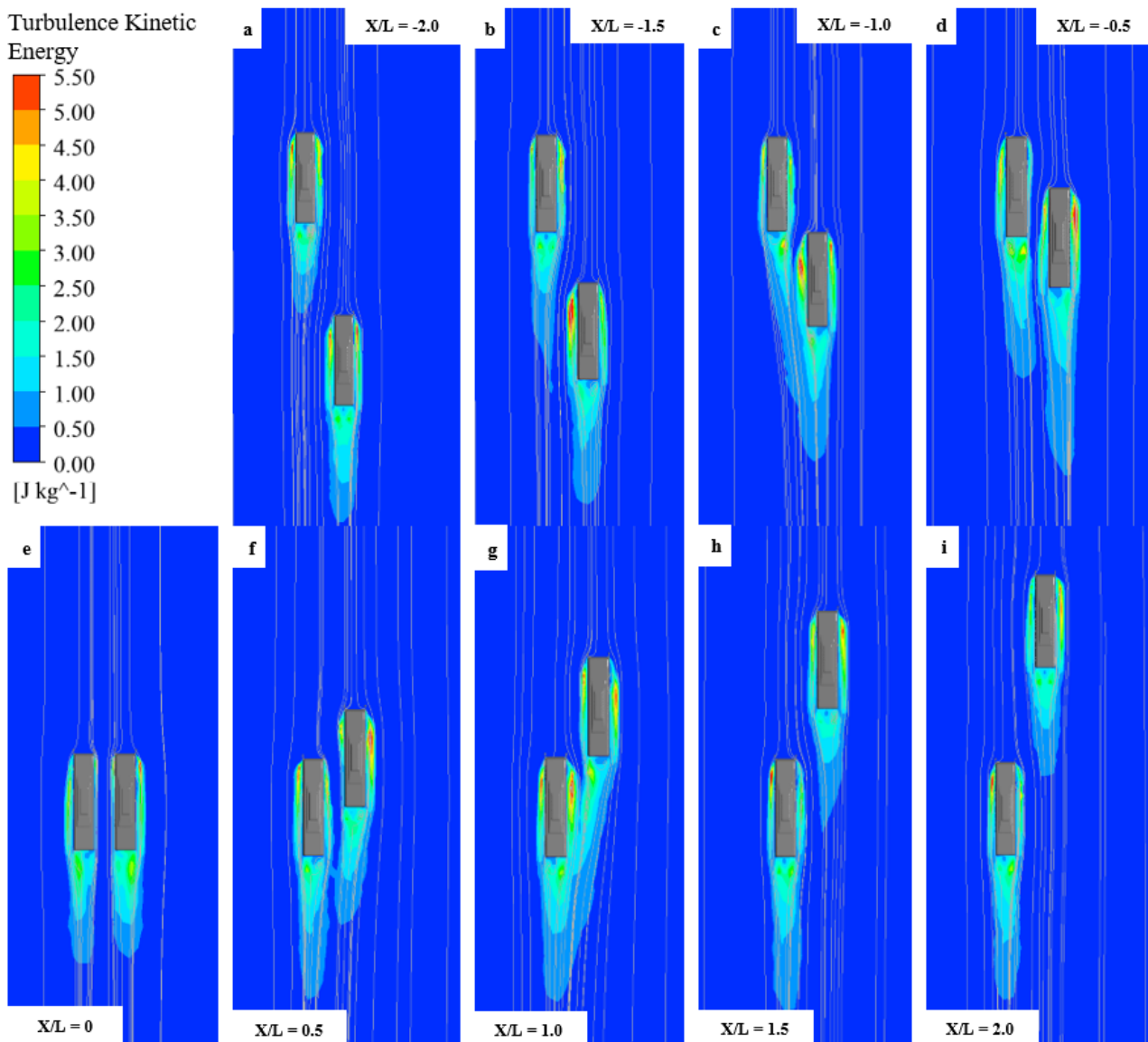
Fig. 5. Effect of  $X/L$  on the aerodynamic force coefficient of Bus1 and Bus2 at different yaw angles

The change of lift coefficient ( $C_L$ ) and side force coefficient ( $C_S$ ) of both buses at the yaw angle between  $0^\circ$  to  $30^\circ$  are depicted in Figure 5(c), Figure 5(d), Figure 5(e), and Figure 5(f) respectively. The value of  $C_L$  and  $C_S$  for bus1 does not change significantly during the overtaking process except when the yaw angle is  $30^\circ$ . However, Bus2 experiences significant value alteration of  $C_L$  and  $C_S$ . When Bus2 is overtaking bus1 at  $X/L = -1$  until  $X/L = -0.5$  the value of  $C_L$  drops close to the value of lift coefficient of the bus without crosswind. A similar trend occurs on the  $C_S$  value of Bus2. When the

yaw angle direction is  $30^\circ$ , the value of  $C_s$  decrease from about 2.7 to 1.5 at  $X/L=-1$  until  $X/L=-0.5$ . It is then increased to the value of about 4.0 at  $X/L=1$ . This can be generally said that the increment of yaw angle increases the value of all aerodynamic coefficients for both busses yet having less significant alteration for Bus1. However, the variation of aerodynamic coefficient differs significantly for Bus2 and results in the peak when Bus2 is overtaking Bus1 at  $X/L=1$ .

### 3.2 Turbulence Kinetic Energy with $\alpha=0$ and $\alpha=30$

Figure 6 depicts the energy produced by air turbulence when Bus2 is overtaking Bus1 with the absence of crosswind. A relatively high value of turbulence kinetic energy occurs unpredictably for two busses. The turbulence kinetic energy is observed through a middle horizontal plane of the bus model. When the  $X/L=-1.5$ , quite high turbulence is observed on the left side of Bus2, then the significant turbulence shifts to the right side of Bus2 when  $X/L=-0.5$ .



**Fig. 6.** Turbulence kinetic energy during overtaking without crosswind

The value is not significant when two buses are in the same row. It is followed by a higher value on both the right side of Bus2 and the left side of Bus1 when  $X/L=1.0$ . Then the kinetic energy is



getting lower towards  $X/L=2.0$ . Another point to be noted is that, when  $X/L$  equals  $-1.0$  and  $1.0$ , the tail of turbulence seems to merge. The separation of turbulence kinetic energy area then started to separate at  $X/L=-0.5$  and  $X/L=1.5$ .

The most severe condition when the crosswind produces  $30^\circ$  of yaw angle is shown in Figure 7. The air stream from the leeward side generates a much more significant effect on the turbulence kinetic energy during the overtaking process. Since the air flows over the bus that is not in the same direction as the bus motion, a wider area generation of turbulence kinetic energy is observed. This area is the generation of wake due to the airflow through the bus that depends on the vehicle's profile and generally becomes larger as the yaw angle increases [30-33]. When  $X/L=-2.0$ , both buses experience a high turbulence kinetic energy on the right side of the bus. As Bus2 is approaching Bus1 both generated turbulence merges altogether and produces a higher turbulence area. When two buses are in the same row, Bus1 does not generate turbulence as much as Bus2. However, when  $X/L=0.5$  Bus1 produces the highest value of turbulence kinetic energy over every position in the overtaking process at the right side of the bus. Since then, the turbulence kinetic energy area is getting separated until Bus1 is completely overtaken by Bus2.

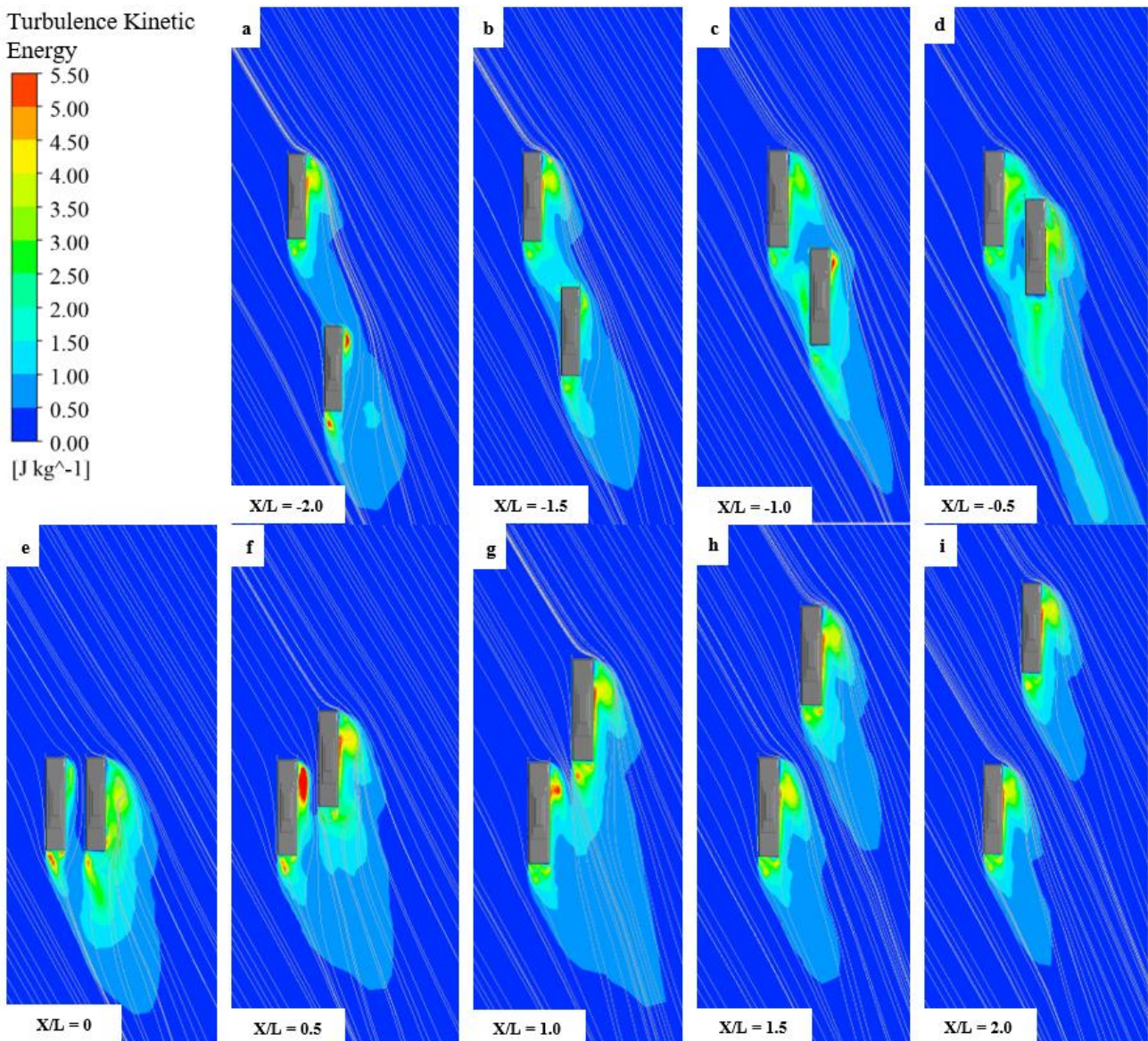


Fig. 7. Turbulence kinetic energy during overtaking at  $\alpha=30$

### 3.3 Pressure Coefficient at $X/L=1$

It has been mentioned in the previous section that the most prominent alteration value of the aerodynamic coefficient occurs when  $X/L=1$ . Figure 8 portrays the pressure coefficient at the bus surfaces and Figure 9 depicts the results of the pressure coefficient at the area around the vehicle at different yaw angle conditions. A positive value of the pressure coefficient can be observed at the front part of the bus when the crosswind is absent. A wide area of the front shield produces the highest positive result in pressure coefficient. However, the increasing value of the yaw angle causes the positive value of the pressure coefficient to shift at the front left part of the bus as depicted in Figure 8. The results of the pressure coefficient area can also be identified on the area around the bus as shown in Figure 9. The positive value of the pressure coefficient occurs on the front part of the bus and is getting shifted to the left side of the bus as the yaw angle elevates. The more yaw angle applied, the results of negative pressure coefficient shift to the left side of the bus. It can be noticed that the negative value of the pressure coefficient occurs at the whirlpool of the streamline of the wind.

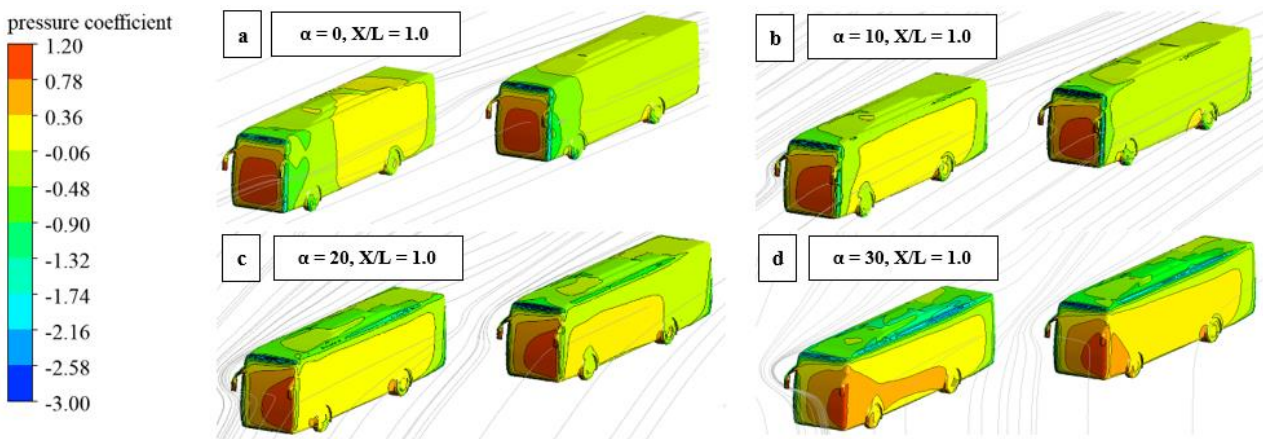


Fig. 8. Pressure coefficient at the bus surface at different yaw angles

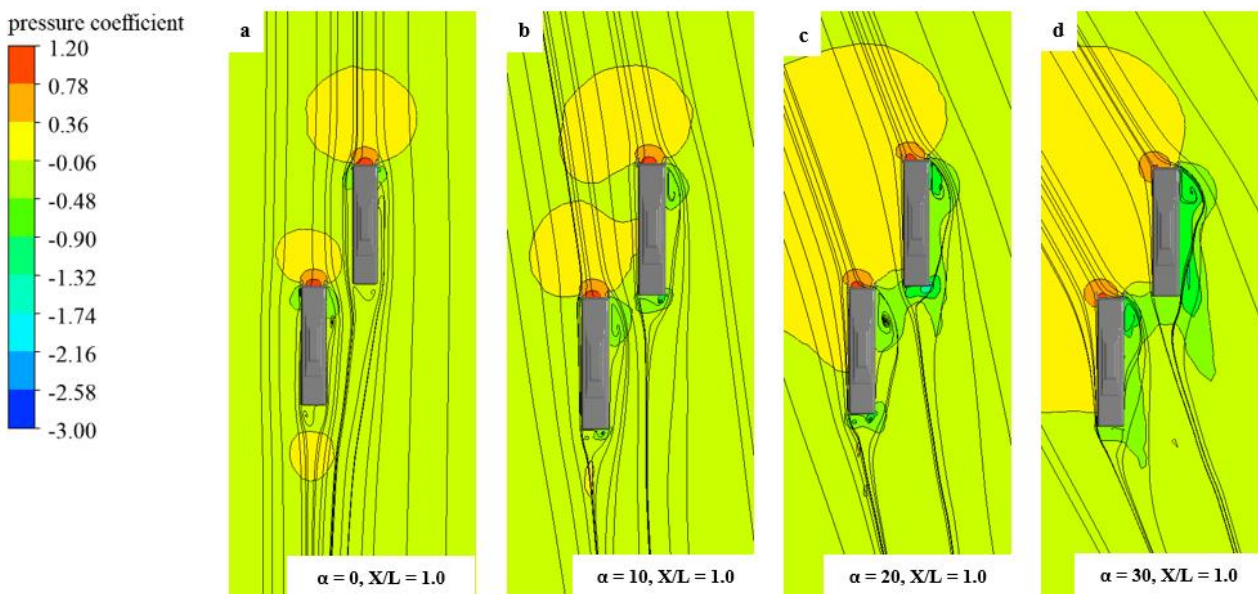
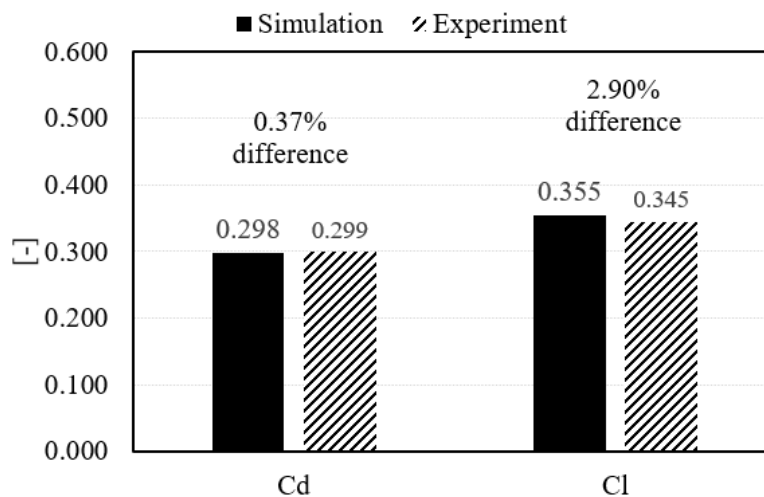


Fig. 9. Pressure coefficient around the vehicle at different yaw angles

#### 4. Results Validation

The accuracy of the calculation of the numerical results needs to be verified. The numerical results validation utilized the comparison between the simulation resulting from the Ahmed body shape and the data obtained through the experiment by Meile *et al.*, [34]. The results on drag coefficient ( $C_D$ ) and lift coefficient ( $C_L$ ) were observed and compared between simulation and experiment. It can be observed that the results of simulation were generally consistent compared to the experimental testing. The different percentage of drag coefficient was about 0.37% and the lift coefficient was 2.90%. Therefore, the simulation results in this study are generally represent the value of the real testing.



**Fig. 10.** Percentage of difference between the simulation and experimental validation

#### 5. Conclusion

The overtaking process of a scaled bus model under crosswind was numerically investigated in a simulation in this study. The aerodynamic coefficient of the bus model changed significantly at the position of  $X/L=1$  as compared to other positions. The presents of crosswind resulted in a higher value of the aerodynamic coefficient for each bus model. The examination in the turbulence kinetic energy showed that a significant increase of turbulence kinetic energy occurred for higher yaw angles particularly at  $X/L=0.5$  and  $XL=1$ . In terms of pressure coefficient, the positive value area became larger as the yaw angle increase giving more whirlpool at the leeward side of the bus. The planned case for further study is to experimentally investigate the influence of the main part of the bus model to study the resulting behavior of the whirlpool around the bus and to increase the aerodynamic performance of the proposed bus model.

#### Acknowledgement

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