



A Numerical Prediction of Stabilized Turbulent Partially Premixed Flames Using Ammonia/Hydrogen Mixture

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ABSTRACT

The objective of this work is to computationally assess the performance of a carbon free ammonia-hydrogen mixture when burnt in a gas turbine like combustor. Recently, utilizing ammonia as an alternative carbon-free fuel for future power, industry applications and achieving clean energy attracted enormous interest. Pure ammonia oxidation is facing many challenges such as high NO_x emissions, high ignition energy, slow reactivity and lower laminar flame speeds. Therefore, the use of ammonia/hydrogen mixture provides flame stability and increasing flame speed. In this manuscript a numerical study for a new swirl stabilized combustor for oxidizing ammonia/hydrogen mixture. Numerical two dimensional model simulations of a turbulent flame on Reynolds Averaged Navier Stokes (RANS) including a realizable k- ϵ turbulent scheme with the aid of chemistry mechanism were performed under various conditions. Partially premixed combustion model with flame-let concept was selected and radiation effects are also considered. Validation for the predicted results showed a reasonable agreement when validated with the experimental data. The results discuss the influence of changing inlet pressure and equivalence ratio on the stability and the characteristics of unburnt NH₃ and NO emissions. Results show that for constant operating conditions such as constant equivalence ratio of 0.8 that increasing hydrogen content resulted in increasing NO emission. Also, for constant ammonia/hydrogen concentrations, NO emissions increases with equivalence ratio then reduced at rich conditions and NH₃ emissions are generally low. Equivalence ratio lower than 1.2 will be preferable to reduce the amount of unburnt NH₃ formation.

1. Introduction

As climate change policies as well as the reduction of capital cost for renewable energy sources for generating power including wind, PV CSP including attempts to utilize the energy content in for example rice husks, [1], it is very unlikely that gas turbine power plants get phased out. Such plants have a long life that range between twenty to forty years, as well as their operation versatility in adding up to the power supply to meet the peak values of the demand. In addition to, utilization of fossil fuels lead to harmful effects to the environment such as greenhouse gases, thus sustainable

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energy resources should be increasingly utilized to avoid these effects. Ammonia (NH_3) is considered as carbon-free alternative fuel having a good potential for future power generation and industry. In addition, it is carrying 17.8% of its weight hydrogen and the products of its complete firing are water vapor and nitrogen only. In comparison, NH_3 has some appealing characteristics relative to hydrogen, such as lower cost for storage, higher volumetric density that can be easily transported, stored in liquid phase, experience in manufacturing and handling come from its history in some industries (e.g. chemical industry), and much safer transport due to its lower reactivity when compared to hydrogen [2]. However, utilizing ammonia in various applications faces some challenges concerning high ignition energy required, lower burning velocity compared to hydrocarbon fuels, narrow flammability ranges and high NO emissions generated during its oxidation [3,4]. However, the problems of narrow flammability ranges and high ignition energy is discussed in studies that focused on improvement of combustion performance of application that utilize ammonia as fuel such as combustion in spark ignition and compression ignition engines [5-10] founded that mixing ammonia with other fuels such as hydrogen enhance reactivity and increase combustion velocity.

Furthermore, experimental and numerical studies investigated the influence of swirling of entrained ammonia mixture on the flame stability under various conditions [11]. Somarathne *et al.*, [12,13] investigated a LES numerical study for combustion of turbulent premixed ammonia/air flames. The results showed that introducing recirculation zones increases mixing and flame stability of the flame under different turbulent intensities and initial temperatures. Moreover, these studies predicted a reduction of NO and unbrnt NH_3 emissions together at specific rich flame conditions (Φ of 1.2 and 1.15) and high initial temperature.

In addition, Valera-Medina *et al.*, [14] presented a one dimensional experimental study for oxidation of ammonia/methane mixture (0.61 NH_3 and 0.39 CH_4) in a swirl stabilized flame at various conditions of pressure (atmospheric and 0.2 MPa) and equivalence ratios. The study showed that at rich conditions ($\Phi=1.2$) there is a clear reduction occurred in NO emissions but emissions of unburnt NH_3 were not correctly investigated. Also, the influence of increasing pressure over atmospheric pressure indicated a reduction in NO emissions. Furthermore, Kurata *et al.*, [15,16] studied a swirl stabilized non premixed flame of NH_3 /air mixture in a 50 kW gas turbine combustor and the study reported a high NO emissions generated compared with environmental requirement of 70 ppm [17]. Other research groups also published studies of comprehensive analysis about emission created from burning ammonia. The influence of increasing pressure over atmospheric pressure in the reduction of NO emissions is reported in the analysis study of Duynslaegher *et al.*, [18], and Hayakawa *et al.*, [19]. However, increasing the pressure was also shown to play an important role in decreasing the burning velocity in NH_3 /air flames [20].

On the other hand, utilizing a detailed chemical kinetic mechanism for ammonia is important to ensure accurate simulation from the combustion model. Chemical mechanisms have been established to model the oxidation of ammonia. Konnov [21] proposed detailed mechanism of ammonia hydrocarbon with 129 species and 1231 steps. Tian *et al.*, [22] presented chemical mechanism for modeling oxidation of premixed ammonia/methane with 84 species and 703 steps. Another mechanism have been studied by Mendiara and Glarborg [23] for oxidation of NH_3/CH_4 with 97 species and 779 steps.

Moreover, modifications for ammonia combustion mechanisms are studied by researchers to be available for various burning conditions. Duynslaegher *et al.*, [24] added reactions to N_2O formation to the mechanism of Konnov. The role of HCN in formation of NO in oxidation hydrocarbon and ammonia-hydrocarbon mixture have been studied by Dagaut *et al.*, [25] via mechanism containing 69 species and 462 reactions. Song *et al.*, [26] utilized a modified reaction mechanism of Klippenstein

et al., [27] to study the oxidation of ammonia at high pressure (30-100 bar) and temperature (450-925K) with mechanism containing 34 species and 204 reactions.

Recently researchers introduced chemical mechanisms for the oxidation of ammonia for utilization ammonia in application as a fuel. Okafor *et al.*, [28] presented mixture of ammonia-methane combustion mechanism, with 59 species and 356 steps, to study the laminar flame speed, with aid of the GRI Mech 3.0 mechanism [29] and that of Tian *et al.*, [22]. Otomo *et al.*, [30] updated the mechanism of Song *et al.*, [26], especially in reactions related to NH_2 , HNO and N_2H_2 and introduced a mechanism with 59 species and 356 reactions, for ammonia and ammonia/hydrogen oxidation. Recently detailed mechanism of Glarborg *et al.*, [31] with 151 species and 1397 steps, describing NO formation through $\text{NNH}/\text{N}_2\text{O}$, prompt and thermal mechanisms. Unfortunately, superimposing CFD analysis with detailed chemistry mechanisms in simulating of combustion model is more expensive numerical computation. Reduced mechanisms are developed as a compromise to this problem. In the past, Miller *et al.*, [32] introduced a reduced mechanism for NH_3 oxidation, with 23 species and 98 reactions. Okafor *et al.*, [33] reduced the detailed mechanism to 130 reactions and 42 species. In addition to, Xiao *et al.*, [34] presented more than reduced mechanisms for oxidation of ammonia/hydrogen and ammonia/methane. The reduction of NO in burning hydrogen in the environment of compression ignition energy was studied by Taib *et al.*, [35].

This present study that investigates a numerical study for a new swirl stabilized combustor for oxidation of ammonia/hydrogen mixture. CFD two dimensional model simulations of a turbulent flame on RANS including a realizable $k-\epsilon$ turbulent scheme with the aid of reduced chemistry mechanism [36] were performed under various conditions. Partial premixed combustion model with flame-let concept was selected. Validation for the predicted results with experimental data [37] showed a great agreement when compared with the measured experiments. Moreover, the influence of changing various parameters on emission characteristics is also discussed.

2. Mathematical Model

The numerical simulation model via ANSYS R19 [38] solves the governing equations in the axisymmetric cylindrical coordinate (r and x), which control the motion direction of the flow in time and space. The general form of the conservation equations are presented and solved as follows

2.1 Governing Equations

Mass conservation equation (Continuity)

$$\frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x_i} (\rho u_i) = 0 \quad (1)$$

Momentum conservation equations

$$\frac{\partial}{\partial t} (\rho u_i) + \frac{\partial}{\partial x_j} (\rho u_i u_j) = -\frac{\partial P}{\partial x_i} + \frac{\partial}{\partial x_j} (\bar{\tau}) - \frac{\partial}{\partial x_j} (\rho \overline{u'_i u'_j}) \quad (2)$$

where, $\frac{\partial P}{\partial x_i}$ is the pressure gradient, $\bar{\tau}$ is the stress tensor.

The stress tensor is given by

$$\bar{\tau} = \mu \left[\left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) - \frac{2}{3} \delta_{ij} \frac{\partial u_i}{\partial x_j} \right] \quad (3)$$

where μ is the molecular viscosity and the volume dilation is expressed in the second term on the right hand side.

Energy conservation equation

$$\frac{\partial}{\partial t} (\rho E) + \frac{\partial}{\partial x_i} (u_i (\rho E + p)) = \frac{\partial}{\partial x_i} (\lambda_{\text{eff}} \frac{\partial T}{\partial x_i} - \sum_k h_k \vec{J}_k + (\bar{\tau}_{\text{eff}} u_i)) + S_h \quad (4)$$

$$E = h_s - \frac{p}{\rho} + \frac{v^2}{2} \quad (5)$$

$$h_s = \sum_j Y_j h_j \quad (6)$$

$$h_j = \int_{T_{\text{ref}}}^T C_p j dT \quad (7)$$

where, E is the total energy, h_s is the sensible enthalpy, λ_{eff} is the turbulent thermal conductivity according to the defined turbulence model, $\lambda_{\text{eff}} \nabla T$ represents energy transfer due to conduction, $\sum_k h_k \vec{J}_k$ represents energy transfer due to species diffusion, $(\bar{\tau}_{\text{eff}} u_i)$ describes the energy transfer due to viscous dissipation and S_h is the energy source term due to chemical reaction and any other heat source. The source term due to chemical reaction is determined as follows

$$S_h = - \sum_j \frac{h_{j0}}{M_{w_j}} R_j \quad (8)$$

where h_{j0} is the enthalpy of formation of species j and R_j is the rate of species production.

2.2 Turbulence Modeling

Realizable k- ϵ model

$$\frac{\partial}{\partial t} (\rho k) + \frac{\partial}{\partial x_j} (\rho k u_j) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] + G_k + G_b - \rho \epsilon - Y_M + S_k \quad (9)$$

$$\frac{\partial}{\partial t} (\rho \epsilon) + \frac{\partial}{\partial x_j} (\rho \epsilon u_j) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_\epsilon} \right) \frac{\partial \epsilon}{\partial x_j} \right] + \rho C_1 S \epsilon - \rho C_2 \frac{\epsilon^2}{k + \sqrt{\nu \epsilon}} + C_{1\epsilon} \frac{\epsilon}{k} C_{3\epsilon} G_b + S_\epsilon \quad (10)$$

where $C_1 = \max \left(0.43, \frac{\eta}{\eta + 5} \right)$, $\eta = S \frac{k}{\epsilon}$, $S = \sqrt{2 S_{ij} S_{ij}}$

The definition of the eddy viscosity is $\mu_t = \rho C_\mu \frac{k^2}{\epsilon}$ while C_μ is variable and calculated as follows

$$C_\mu = \frac{1}{A_0 + A_s \frac{k U^*}{\epsilon}} \quad (11)$$

$$\text{where } U^* \equiv \sqrt{S_{ij}S_{ij} + \tilde{\Omega}_{ij}\tilde{\Omega}_{ij}}$$

$$\text{And } \tilde{\Omega}_{ij} = \Omega_{ij} - 2 \varepsilon_{ijk}\omega_k$$

$$\Omega_{ij} = \overline{\Omega}_{ij} - \varepsilon_{ijk}\omega_k$$

where $\overline{\Omega}_{ij}$ is the mean rate of rotation tensor in a frame rotating with the angular velocity ω_k .

The constants A_0 and A_s are given by

$$A_0 = 4.04, A_s = \sqrt{6} \cos \phi$$

where these variables are illustrated in ANSYS R19 [38] as follow

$$\phi = \frac{1}{3} \cos^{-1}(\sqrt{6}W), W = \frac{S_{ij}S_{jk}S_{ki}}{\tilde{S}^3},$$

$$\tilde{S} = \sqrt{S_{ij}S_{ij}}, S_{ij} = \frac{1}{2} \left(\frac{\partial u_j}{\partial x_i} + \frac{\partial u_i}{\partial x_j} \right)$$

2.3 Combustion Modeling

Partial premixed model has been selected to predict the flame characteristic due to the reactivity difference between fuel mixtures as reported in the laminar flame speed results shown in Otomo *et al.*, [30]. The probability density function (PDF) model is preferred in simulating the influence of chemical kinetic in turbulent flows with partially premixed combustion model [39]. The following transport conservation equations are considered to integrate chemical reactions of the kinetic mechanism [36] into turbulence calculations. The model solves a transport equation for the mean reaction progress variable, c (to determine the position of the flame front),

$$\frac{\partial}{\partial t}(\rho c) + \nabla \cdot (\rho \vec{v} c) = \nabla \cdot \left(\frac{\mu_t}{S_{ct}} \nabla c \right) + \rho S_c \quad (12)$$

where, c is the mean progress variable, S_{ct} is the turbulent Schmidt number, \vec{v} is the velocity vector and S_c is the source term of reaction progress. Moreover, the model solves a transport equation for the mean mixture fraction, \bar{f} and the mixture fraction variance, $\overline{f'^2}$. The Favre mean (density-averaged) mixture fraction equation is

$$\frac{\partial}{\partial t}(\rho \bar{f}) + \nabla \cdot (\rho \vec{v} \bar{f}) = \nabla \cdot \left(\frac{\mu_t}{\sigma_t} \nabla \bar{f} \right) \quad (13)$$

There is no source term in Eq. (11) because that elements are conserved in chemical reactions.

$$\frac{\partial}{\partial t}(\rho \overline{f'^2}) + \nabla \cdot (\rho \vec{v} \overline{f'^2}) = \nabla \cdot \left(\frac{\mu_t}{\sigma_t} \nabla \overline{f'^2} \right) + C_g \mu_t (\nabla \bar{f})^2 - C_d \rho \frac{\varepsilon}{k} \overline{f'^2} + S_{user} \quad (14)$$

where $f' = f - \bar{f}$. The default values for the constants σ_t , C_g , and C_d are 0.85, 2.86, and 2.0, respectively, and S_{user} is any user-defined source term. The mean reaction rate in Eq. (10) is modeled as [40]

$$\rho S_c = \rho_u U_t |\nabla_c| \quad (15)$$

where ρ_u is the density of unburnt mixture and U_t is the turbulent flame speed.

$$U_t = A (u')^{\frac{3}{4}} S_l^{\frac{1}{2}} \alpha^{\frac{-1}{4}} l_t^{\frac{-1}{4}} \quad (16)$$

$$U_t = A u' \left(\frac{\tau_t}{\tau_c}\right)^{\frac{1}{2}} \alpha^{\frac{-1}{4}} l_t^{\frac{-1}{4}} \quad (17)$$

A is the model constant, the default value given by Zimont [41] is 0.52 and was later reduced to 0.38 following Yehia & Abdel-Raheem [42], u' is the root mean square velocity, S_l is the laminar flame speed, l_t is turbulence length scale, $\tau_t = l_t/u'$ is the turbulent time scale and $\tau_c = \alpha/S_l^2$ is the chemical time scale.

2.4 Scalar Quantities

Fraction of species and temperature or any other mean scalar is denoted by $\bar{\phi}$, are determined via probability density function (PDF) of f and c as

$$\bar{\phi} = \int_0^1 \int_0^1 \phi(f, c) p(f, c) df dc \quad (18)$$

The probability density function model of β -PDF shape, for the mixture fraction (f) given by

$$p(f) = \frac{f^{\alpha-1} (1-f)^{\beta-1}}{\int f^{\alpha-1} (1-f)^{\beta-1} df} \quad (19)$$

where

$$\alpha = \bar{f} \left[\frac{\bar{f}(1-\bar{f})}{\bar{f}^2} - 1 \right] \quad (20)$$

And

$$\beta = (1 - \bar{f}) \left[\frac{\bar{f}(1-\bar{f})}{\bar{f}^2} - 1 \right] \quad (21)$$

2.5 Kinetic Mechanism

The reduced Okafor's mechanism [36] which reduced from Okafor's mechanism [33] for the oxidation of NH_3/H_2 is selected due to its lower computational cost achieved via using lower number of species in chemistry calculations. The kinetic mechanism containing 19 species and 63 reactions was validated by the author [33] against Okafor's mechanism [36] and experimental data from others [20,43-46] for laminar flame speed calculated. The reduced mechanism [36] is described in supplementary material.

Diagram in Figure 1 presented a formation of NO during oxidation of NH_3/H_2 mixture reaction at stoichiometric condition [36]. Paths showed that the major role of oxidation HNO, NH, NH_2 and N in production of NO. Thickness of the paths illustrates the high and low contribution in production of NO during oxidation of NH_3/H_2 flame. Ammonia is initially produced NH_2 when it consumed with O

and OH and then NH₂ presented NH and HNO when it reacted with OH, H and H. The major amount contribution in production of NO is considered to come from the reactions of HNO with OH, O₂ and H. The reactions of NH with O/O₂ and N with OH/O₂ converted to NO but with lower contribution when compared to HNO.

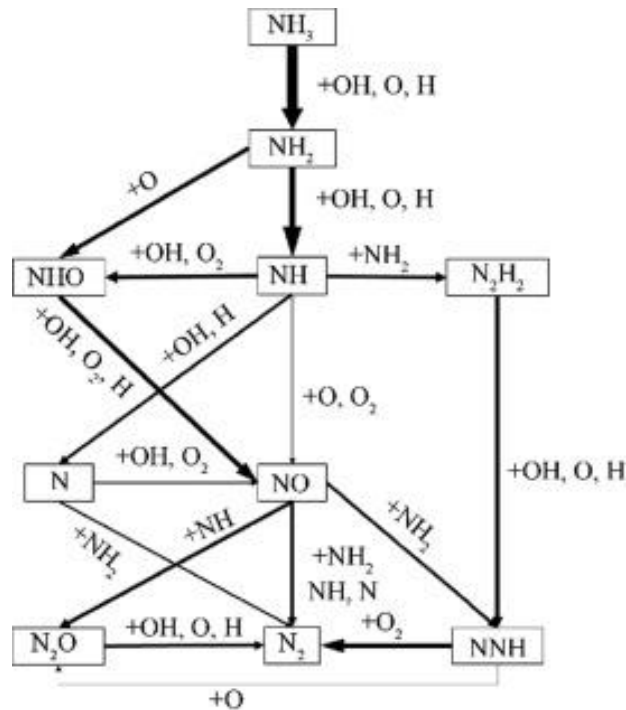


Fig. 1. Path analysis describing the oxidation of NH₃/H₂ flame [36]

The kinetic mechanism [36] is imported into the partially premixed combustion model to simulate oxidation of ammonia/hydrogen mixture exactly. The modeling of chemical reaction in the chemistry mechanism is generated via PDF (Probability Density Function) model. The laminar flame speed have a major role in simulating the model precisely, but the laminar flame speed data available in ANSYS FLUENT [38] is obtained from other research studies for hydrogen, methane, acetylene, ethylene, ethane and propane fuels in various conditions of pressures, temperatures and equivalence ratios [47]. In addition to, the laminar flame speed changes with the change of chemical composition and equivalence ratio. For these reasons, accurate laminar flame speed needs to be calculated from one dimensional prediction study or from experimental measurements. Thus, the data of laminar flame speed variations with equivalence ratio are taken from Otomo *et al.*, [30], due to its account for the various conditions in the present study (Figure 2).

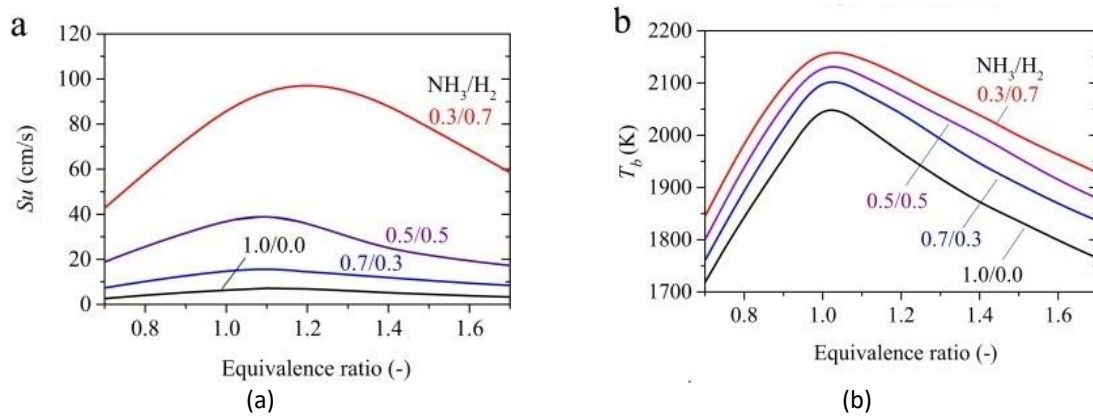


Fig. 2. (a) burning velocity, S_u and (b) adiabatic flame temperature, T_b in NH_3/H_2 /air mixtures [30]

3. Problem Description

The combustor is vertically fired (as shown in Figure 3) with 100 mm diameter and 320 mm height. Initially, air is introduced tangentially inside the premixing tube of 400 mm length, while fuel mixture is issued radially from 12 holes of 2 mm diameter. To attain improved mixing and flame stability, the burner combines air swirl with a bluff body. The 1.9 kW capacity burner shown in Figure 1 utilized in the present study is a swirl-stabilized ammonia/hydrogen mixture used in the study of Costa, *et al.*, [37].

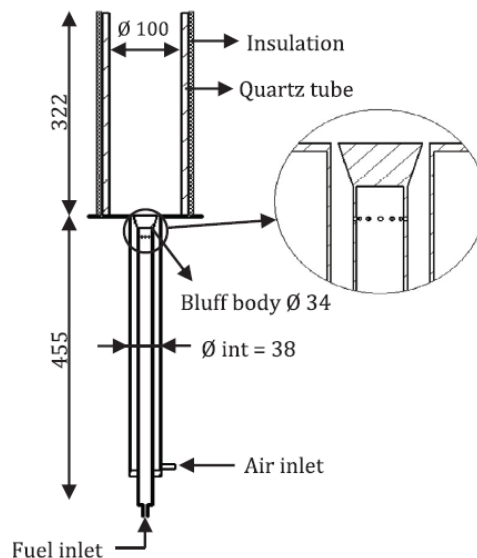


Fig. 3. Combustor geometry [37]

Figure 4 shows 2D axisymmetric geometry for combustion chamber. A axisymmetric for the combustor has been taken for the present study to minimize the simulation time.

The utilizing meshing consists of 3.5×10^5 triangle elements as shown in Figure 3, with quality of (1.56 in average) aspect ratio for total elements. Near burner tip and outlet the mesh required to be condensed due to the importance of the flame in these regions as shown in Figure 5. Figure 6 shows the grid independence study for get agreement for utilizing mesh grid.

The predicted radial temperature at different locations in combustion chamber for case of 80% NH_3 with 0.8 equivalence ratio is compared for different mesh grid sizes of 2.5×10^5 cells, 3.5×10^5 cells and 6.5×10^5 cells. The grid size of 2.5×10^5 cells shows a deviation in predicted radial temperature in

comparison with the grid size of 3.5×10^5 cells and 6.5×10^5 cells which gives results almost identical. Thus, choosing grid size of 3.5×10^5 cells are deemed to be suitable for this study.

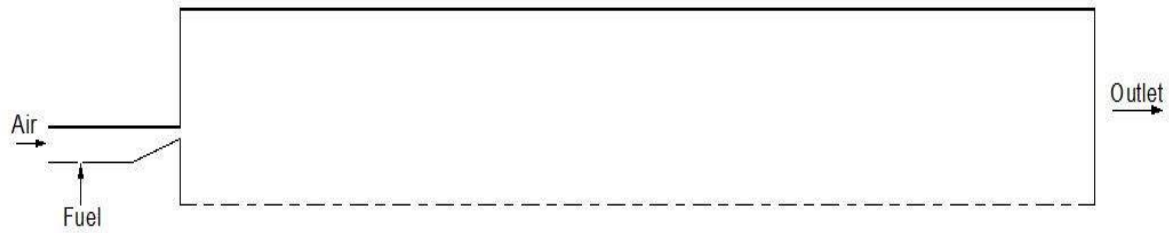


Fig. 4. 2D axisymmetric geometry for combustion chamber

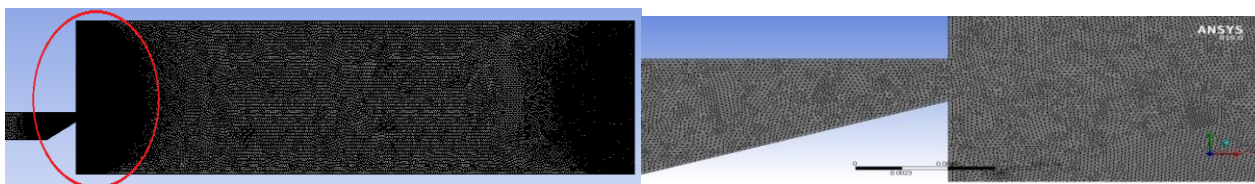


Fig. 5. 2-D geometry mesh

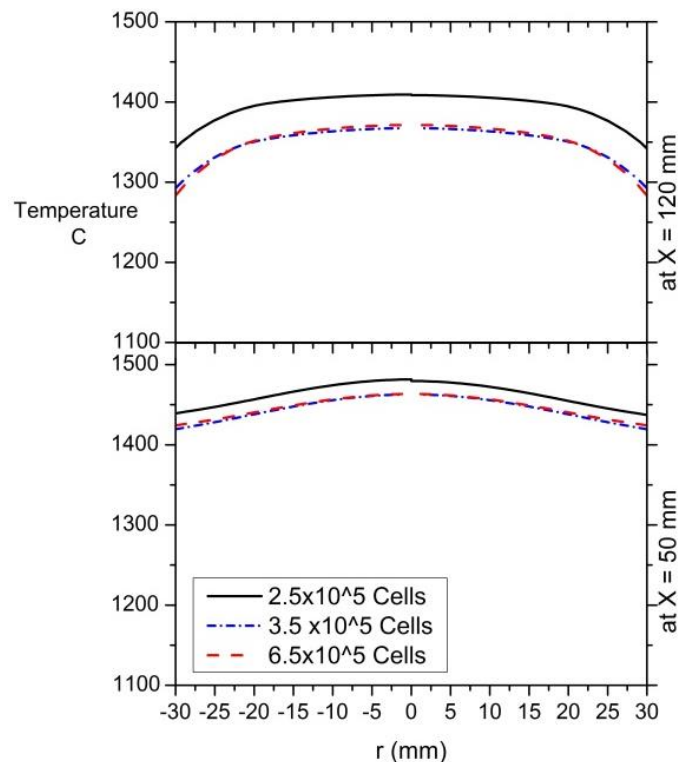


Fig. 6. Radial temperature comparison for various mesh sizes at different locations in the combustor

4. Cases Studied and Boundary Conditions

The 2-d model of combustion is partially premixed model with steady non adiabatic flame-let mixture and the radiation model is do. Simulations of a turbulent flame in the present study rely on Realizable $k-\epsilon$ turbulent scheme due to its ability to deal with swirling flow when compared with other $k-\epsilon$ models and to minimize time cost [48].

Discretization for flow conservation equations is utilized 2nd order upwind scheme and for pressure and velocity coupling the Coupled algorithm is selected. In order to attain accuracy in the predicted results the solution convergence criteria is controlled via monitoring the residual of continuity, momentum, and energy are less than 10^{-3} , 10^{-5} and 10^{-6} , respectively.

The air and fuel enter at 300 K and the walls have convection boundary condition with $10 \text{ W/m}^2\text{K}$ heat transfer coefficient and the stream of air at 300 K except for walls of chambers not the bottom of chamber kept to be coupled boundary condition.

5. Results

The predicted results are firstly validated with experimental data [37] for cases of different concentration of NH_3 and at equivalence ratio as shown in Table 1.

Table 1
Conditions of validated cases

Cases	Concentration of NH_3	Φ	P (Mpa)
Flame 1	70% NH_3	0.8	0.1
Flame 2	80% NH_3	0.8	0.1
Flame 3	90% NH_3	0.7	0.1
Flame 4	90% NH_3	0.8	0.1
Flame 5	90% NH_3	0.9	0.1

The dry volumetric emissions of O_2 and NO_x are validated with the experimental data [37] as shown in Figure 5 and 6 respectively. The axial temperature is validated with the experimental data [37] that shown in Figure 8 to present agreement with experimental one (Figure 7) [37] except in the location of entrained mixture from the annuals due to the increasing of mixing at this locations. Enhancement of the mixture mixing and the stability of the flame in the burner is due to the effect of tangentially entrained air and the introduced bluff body that leads to generation of internal recirculation zone at the combustor region where the temperature increases and then the dilution occurred and the temperature then decreased till to reach outer recirculation zone.

The axial temperature distribution along the axis of the combustor illustrates the increase in the temperature near the bottom of the burner. Therefore, the increase in temperature near the bottom of the burner is due to the influence of recirculation zone in increasing mixing and also indicates that hydrogen concentration is utilized near the bottom of the burner.

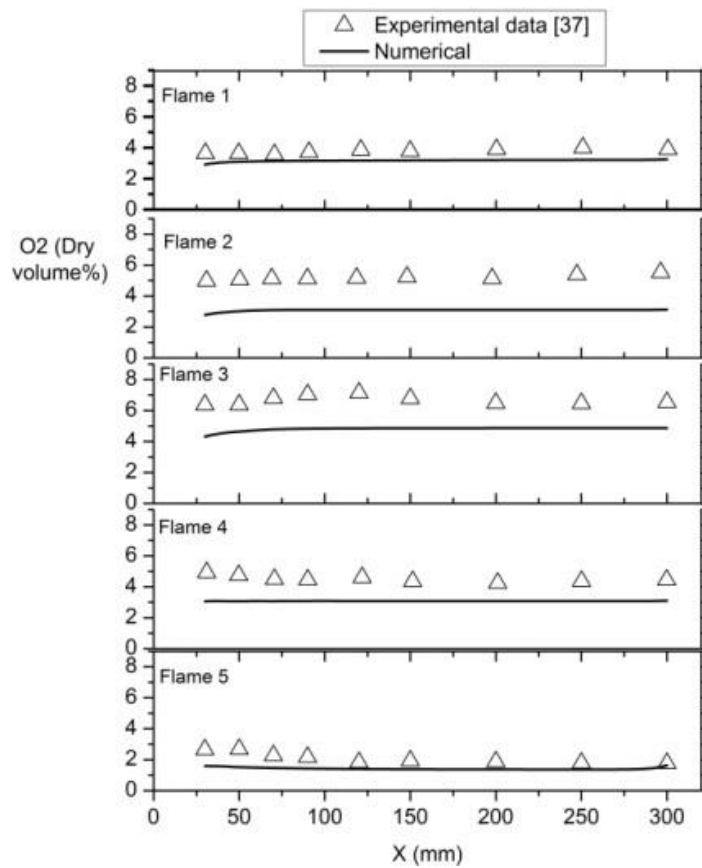


Fig. 7. Comparison of the experimental data [37] and calculated O₂ at the center of the combustion chamber

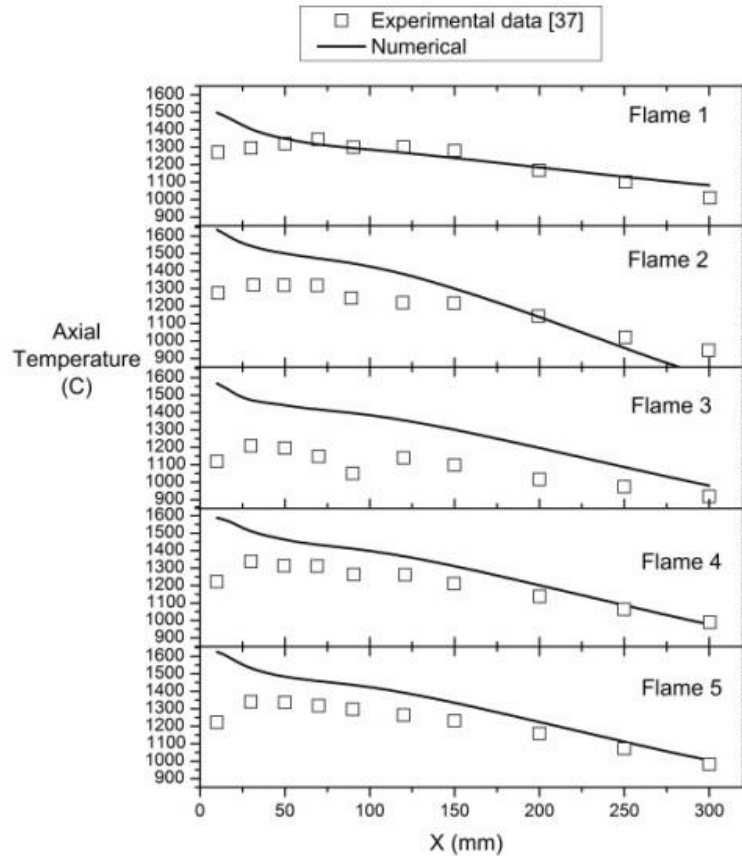


Fig. 8. Comparison of the experimental and calculated axial temperature at the center of the combustor [37]

5.1 Flame Stability

In order to examine the influence of swirling flow that introduced via bluff body burner design and the stability of ammonia/hydrogen flame in the burner, study of flow field and combustion characteristics via numerical simulations are adopted. The flow field formed from the burner is presented in Figure 9. The flow field pattern introduces two recirculation zones, the primary one is inner recirculation zone (IRZ) that is mainly related to the introduced swirling flow and bluff-body design, whereas smaller external recirculating zone (ERZ) between the high velocity zone and the bounding walls, generated by the rapid expansion of the swirling flow from annular into the burner [49,50]. As a result of generation of the IRZ in the burner, the stability of the mixture is achieved by the enhancement of the mixture mixing via carrying the radicals, unburnt and burnt gases into combustion chamber. The reduction in the average velocity developed in the IRZ lead to increasing the time of chemical reaction resulting in relatively higher reaction rates.

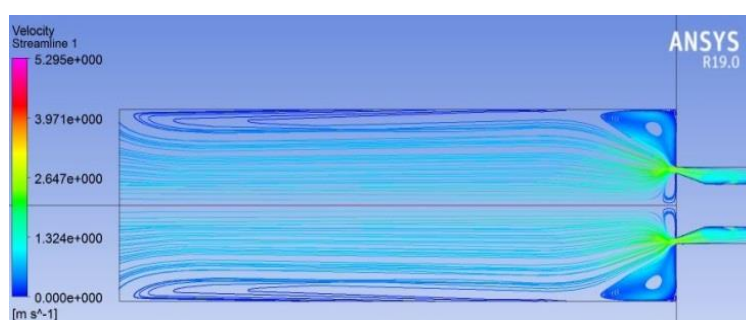


Fig. 9. Velocity streamline for case of 70% NH₃ at $\Phi=0.8$

5.1.1 Flame characteristics at lean condition

Moreover, the present study investigated the influence of mixing ammonia with different compositions of hydrogen at lean condition ($\Phi=0.5$) as shown in Figure 10 (a, b and c). Higher temperature near the burner tip and longer flame length is considered for high hydrogen concentration in the mixture due to high reactivity and burning velocity of hydrogen. However, temperature contours in Figure 10 (d and e) show that the reduction of hydrogen composition in the mixture leads to blow off of the flame into the chamber.

On the other hand, higher activity of hydrogen as compared to pure ammonia oxidation flames leads to flash back occurring into the chamber as shown in Figure 12 (a, b and c). Contours of temperature illustrate that in cases of high hydrogen concentration in the mixture, the increase in equivalence ratio towards to rich condition leads to the flash back into the chamber. While, Figure 12 (d and e) show that the reduction in hydrogen composition in the mixture at increasing of equivalence ratio will not lead to blow off condition to the flame as in cases of equivalence ratio is lower than 0.8.

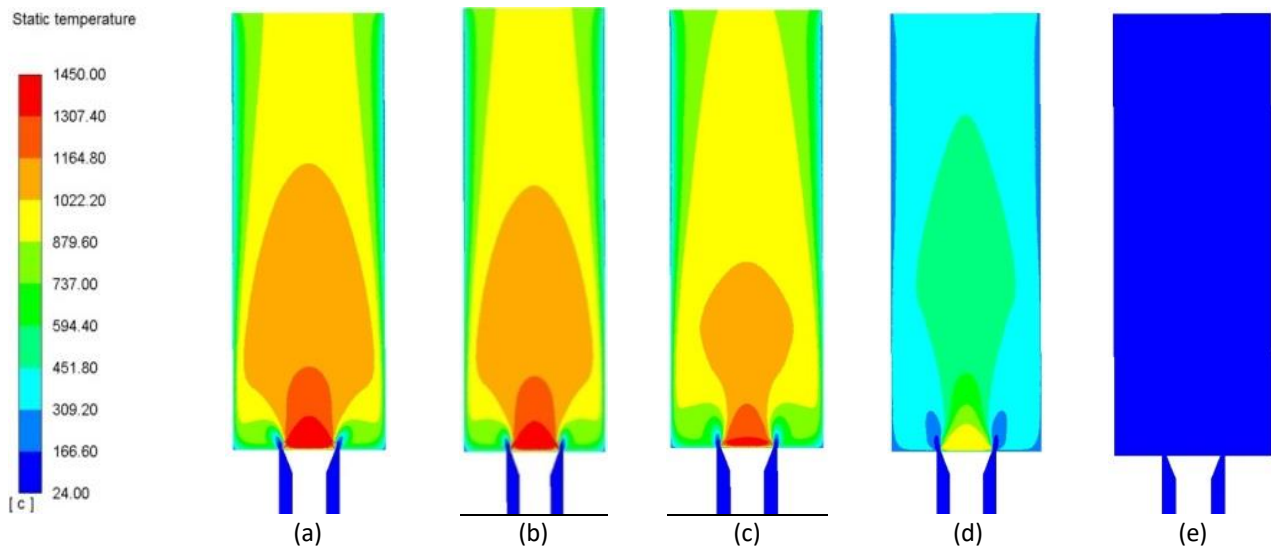


Fig. 10. Temperature contours for different fuel composition (a=50%, b=40% c=20%, d=10% and e=0% H₂) at $\Phi=0.5$

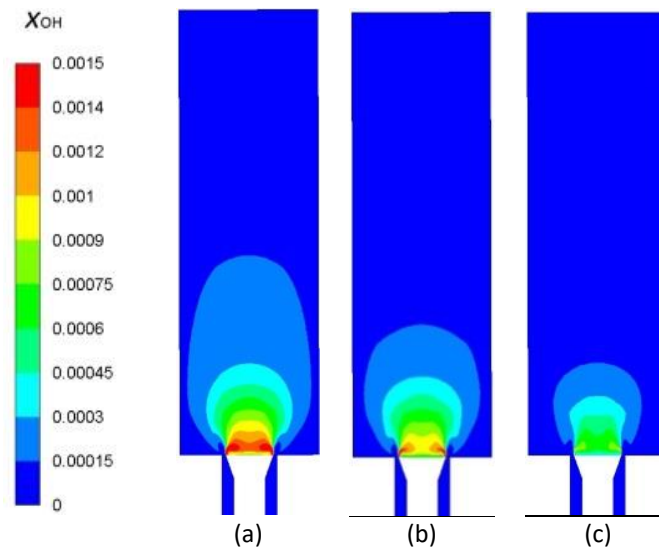


Fig. 11. Mole fraction of OH contours for different fuel composition (a=50%, b=40% and c=20% H₂) $\Phi=0.5$

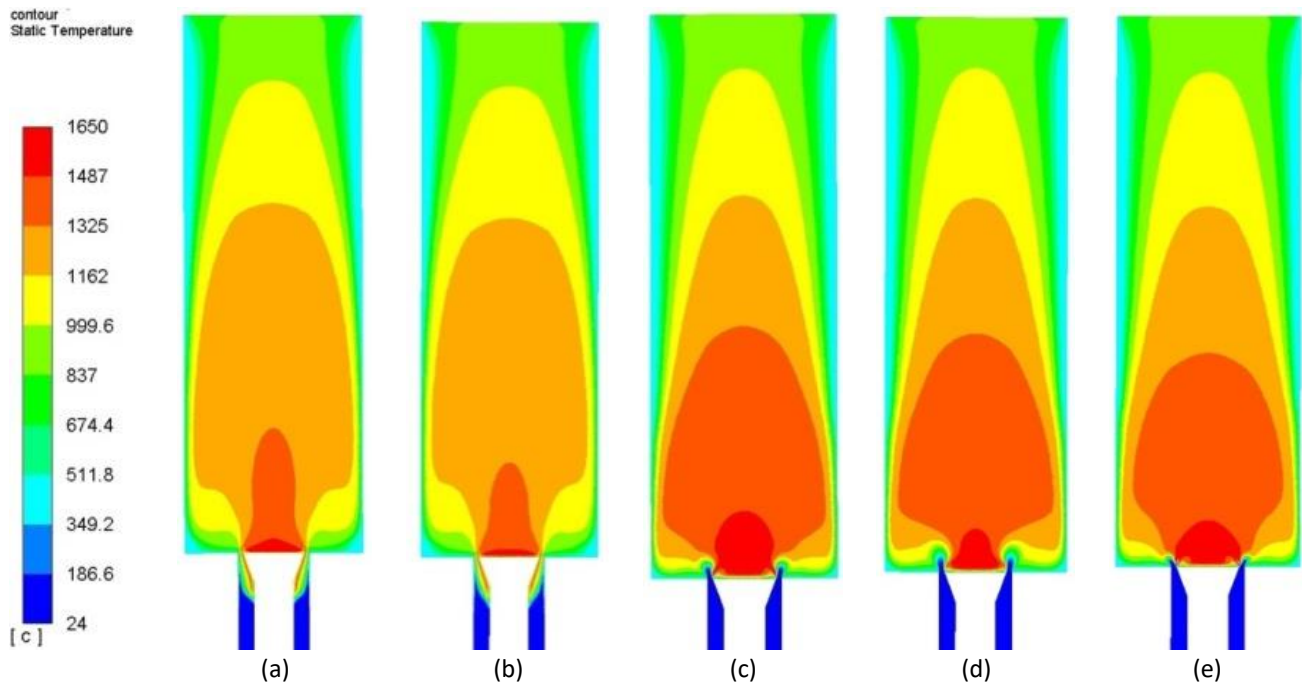


Fig. 12. Temperature contour for cases of (a=50%, b=40% c=20%, d=10% and e=0% H₂) at $\Phi=0.8$

5.1.2 Flame characteristics at stoichiometric

The flame characteristics of NH₃/H₂ mixture under the stoichiometric condition and initial mixture temperature of 300 K are shown in the Figure 13 (a, b) in terms of temperature contour and mole fraction of OH respectively. OH distribution in the chamber illustrates an increase in OH followed by an increase in the temperature. The important influence of changes of OH concentration on emissions NO and unburnt NH₃ will be illustrated in the next section.

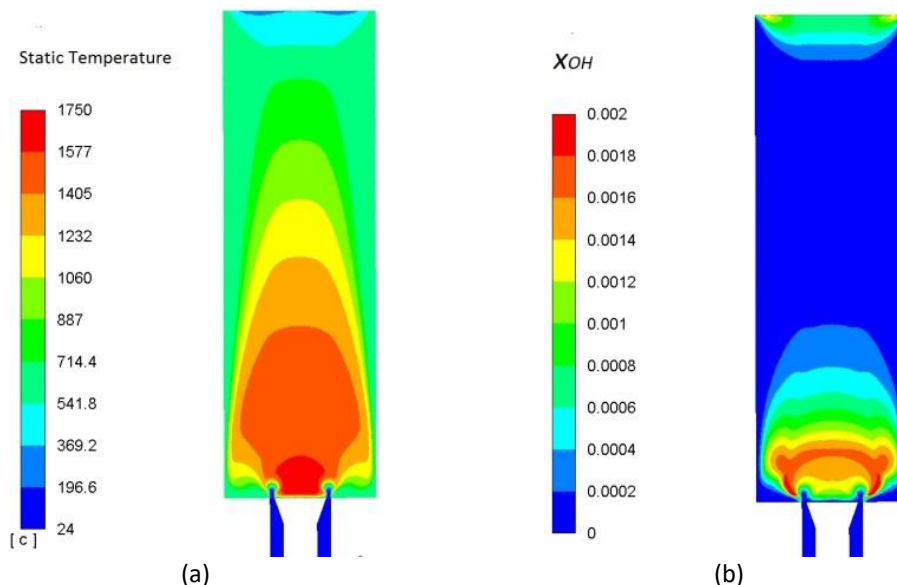


Fig. 13. flame characteristics at 70%NH₃ and stoichiometric condition: (a) Temperature contour; (b) mole fraction of OH

5.1.3 Flame characteristics at rich condition

The flame characteristics of NH_3/H_2 mixture under the rich condition of equivalence ratio equal to 1.2 with initial mixture temperature of 300 K are shown in the Figure 14 (a, b) in terms of temperature contour and mole fraction of OH respectively. OH distribution in the chamber illustrates a reduction in OH concentration as compared with the stoichiometric condition and its effects on decrease of temperature and flame length is shown in Figure 14 (a).

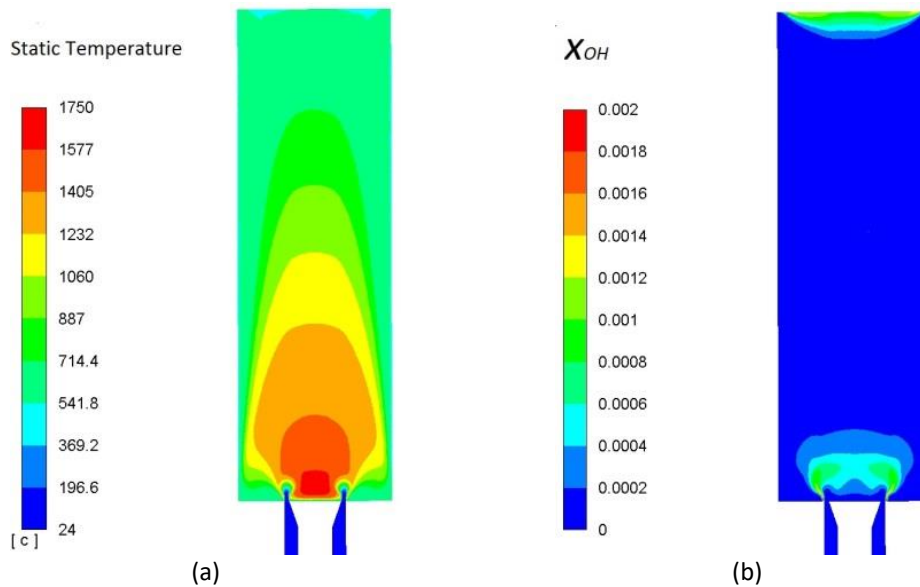


Fig. 14. Flame characteristics at 70% NH_3 at rich condition: (a) Temperature contour; (b) mole fraction of OH

The effect of increasing hydrogen concentration in the mixture on the NO emission is shown in Figure 15. At constant equivalence ratio the increasing in hydrogen content lead to increase of NO_x emission.

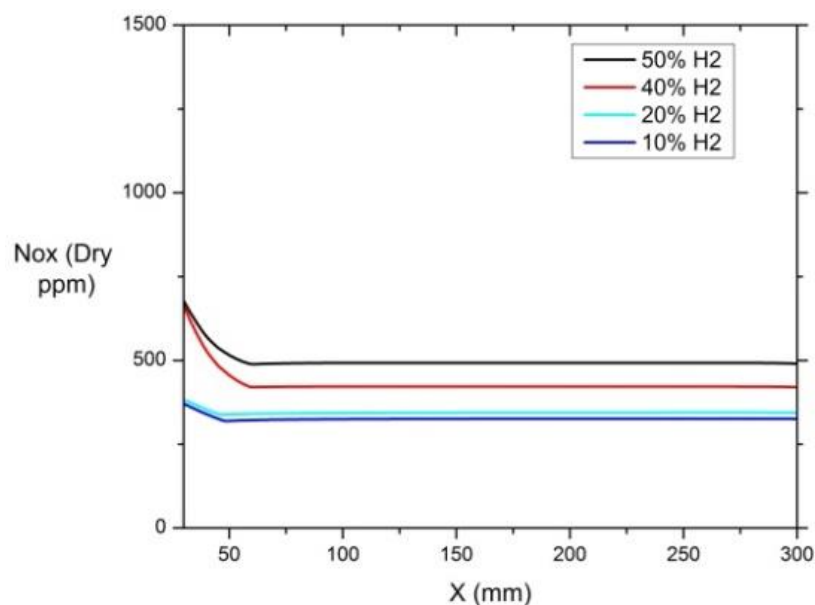


Fig. 15. NO_x formation at different hydrogen composition at 70% NH_3 and $\Phi=0.8$

5.2 Effect of Equivalence Ratio

The emission characteristics for oxidation of NH_3 / H_2 from lean to rich condition is analyzed at inlet mixture temperature of 300 K and atmospheric pressure for 70% NH_3 case to show the effect of changing equivalence ratio. Figure 16 (a, b, c and d) and Figure 17 Show the mole fraction of unburnt NH_3 and NO_x formation respectively.

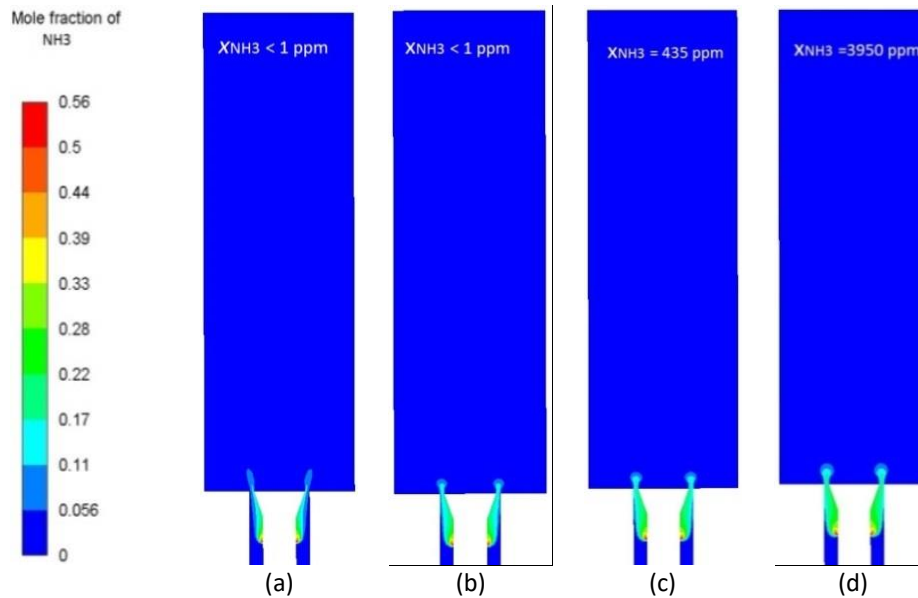


Fig. 16. Effect of changing Φ on mole fraction of NH_3 emission at atmospheric pressure for 70% NH_3 at various equivalence ratio (a=0.5, b=0.8, c=1 and d=1.2)

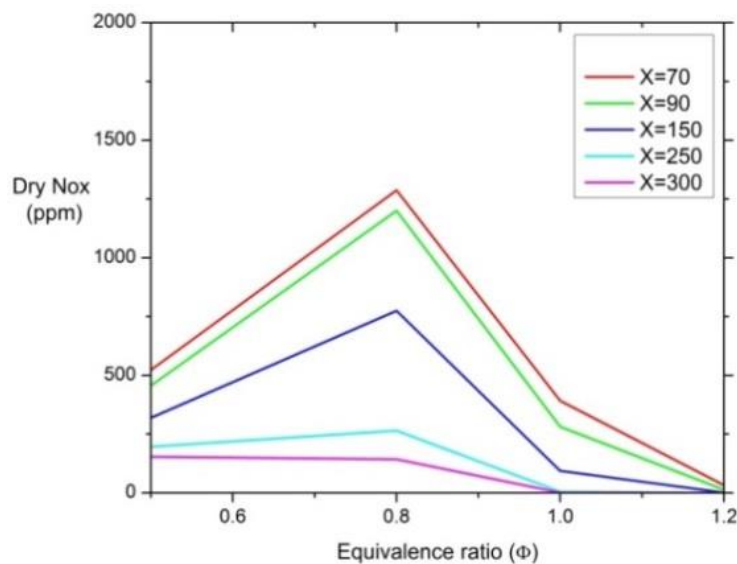


Fig. 17. Effect of changing Φ on NO_x formation at atmospheric pressure and 70% NH_3 predicted at different location in the combustor

The unburnt NH_3 increases with respect to equivalence ratio from lean to rich conditions. At rich condition the amount of unburnt NH_3 show a significant increase, thus equivalence ratio lower than 1.2 will be preferable to reduce the amount of unburnt NH_3 formation. However, the NO emission characteristic increase with the increase of equivalence ratio to its maximum value around $\Phi=0.8$ till

reaches the rich condition and then start to decrease. OH radicals have a major role in the destruction of NO formation. OH concentration illustrated in previous section reported that the reduction in OH radicals faced with decrease in NO emission. The rich conditions have an important influence in the reduction of NO and unburnt NH₃ emissions but the correct value of the equivalence ratio at rich condition have to be selected to achieve the reduction in both of unburnt NH₃ and NO emissions.

5.3 Effect of Pressure

Figure 18 and 19 shows that the increasing of operating mixture pressure over the atmospheric pressure to 0.5 MPa have a significant effect in the reduction of emission concentrations of unburnt NH₃ and NO. The reduction of OH radicals have important role in the reduction of NO formation and unburnt NH₃ [19].

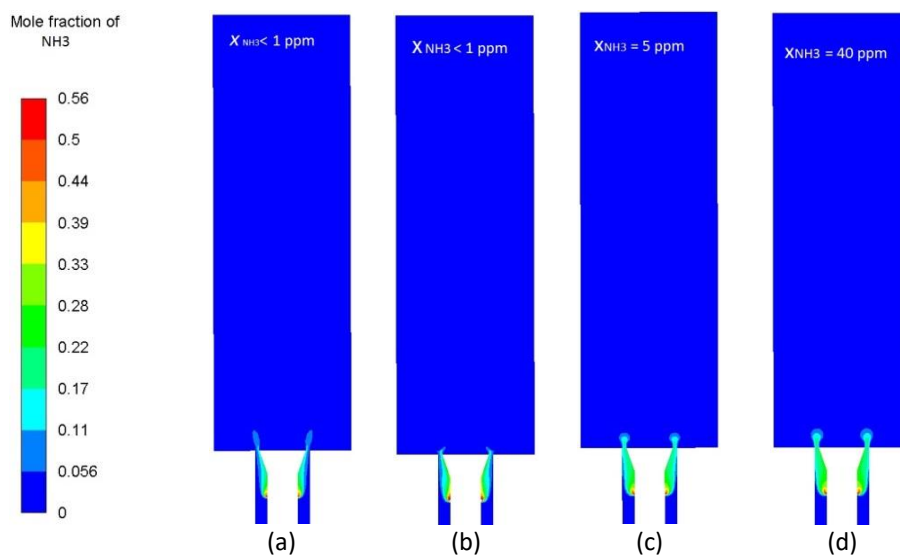


Fig. 18. Effect of changing Φ on mole fraction of NH₃ at P=0.5MPa and 70% NH₃ at various equivalence ratio (a=0.5, b=0.8, c=1 and d=1.2)

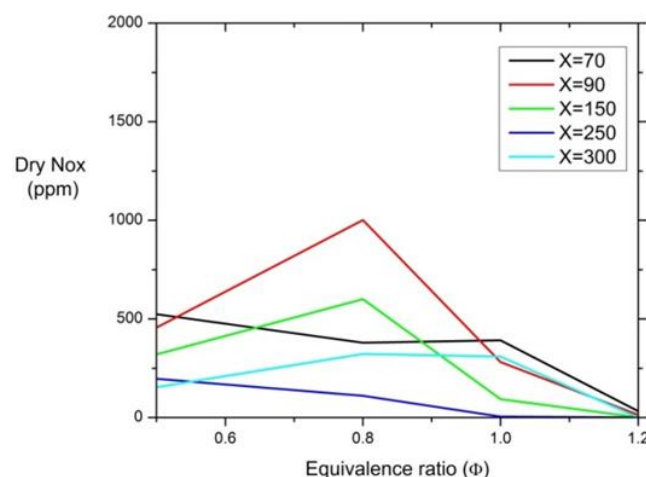


Fig. 19. Effect of changing Φ on mole fraction of NO at P=0.5MPa and 70% NH₃ predicted at different location in the combustor

6. Conclusion

The computational investigation of burning ammonia hydrogen mixtures in a gas turbine like combustor resulted in the following conclusions

- i. As a result of generation of the IRZ in the burner, the stability of the mixture is achieved by the enhancement of the mixture mixing via carrying the radicals, unburnt and burnt gases into combustion chamber
- ii. High hydrogen concentration in the mixture at slightly lean condition leads to the flash back into the chamber
- iii. At lean condition lower than 0.8 equivalence ratio the reduction of hydrogen composition in the mixture leads to blow off of the flame into the chamber
- iv. The rich conditions have an important influence in the reduction of NO and unburnt NH₃ emissions but the correct value of the equivalence ratio at rich condition have to be selected to achieve the reduction in the unburnt NH₃ and NO emissions
- v. Equivalence ratio lower than 1.2 will be preferable to reduce the amount of unburnt NH₃ formation
- vi. The reduction in OH radicals faced with decrease in NO emission
- vii. The increasing of operating mixture pressure to 0.5 MPa have an significant effect in the reduction of emission concentrations of unburnt NH₃ and NO
- viii. For constant condition such as constant $\Phi=0.8$ increasing hydrogen content resulted in increasing NO_x emissions
- ix. For constant ammonia/hydrogen concentrations, NO emissions increases with equivalence ratio then reduced at rich conditions and NH₃ emissions are generally low

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